

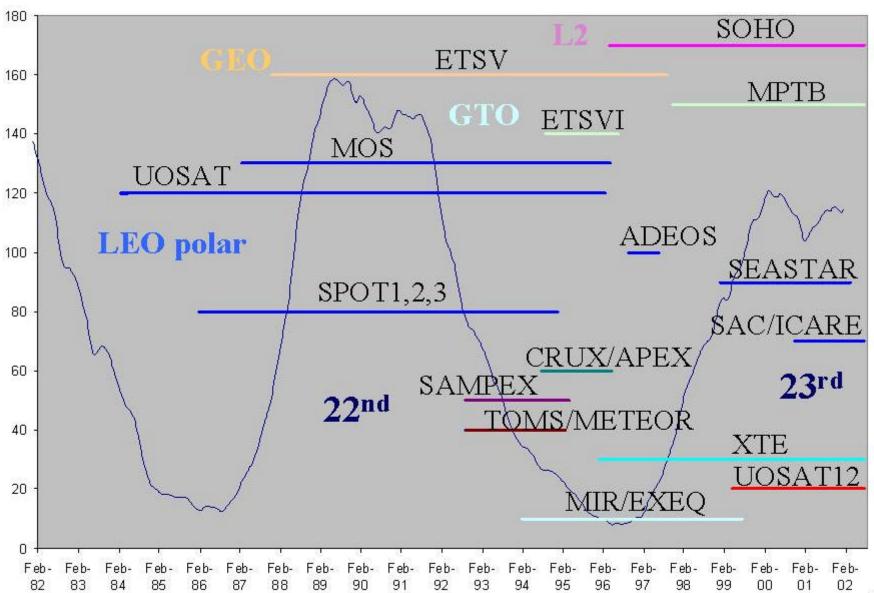
# Lessons Learned from Radiation-induced Effects on Solid State Recorders and Memories

C. Poivey, J. Barth, G. Gee, H. Safren, K. LaBel

## **Outline**

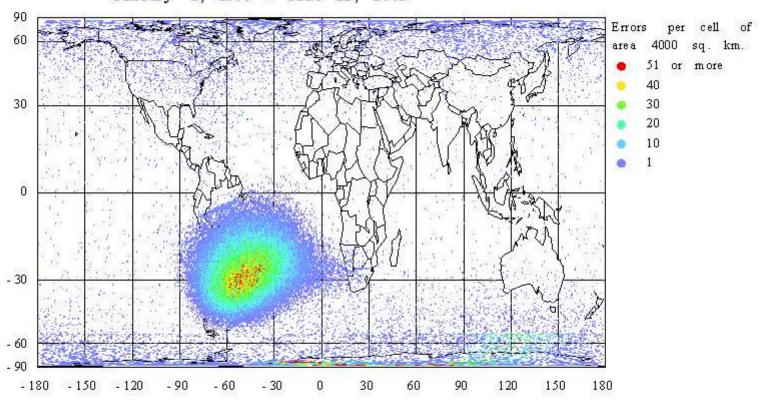
- Introduction
- Examples:SEASTAR,APEX/CRUX
- Accuracy of upset rate calculations
- Efficiency of SEU mitigation schemes
- Conclusion

#### Introduction

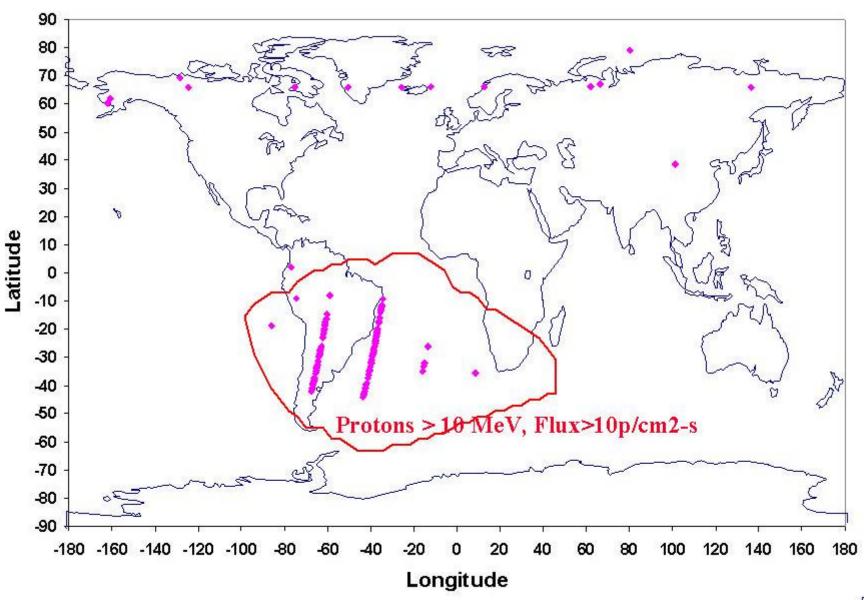


# Flight Data Examples, Seastar-Geographic Plot of Single Hits

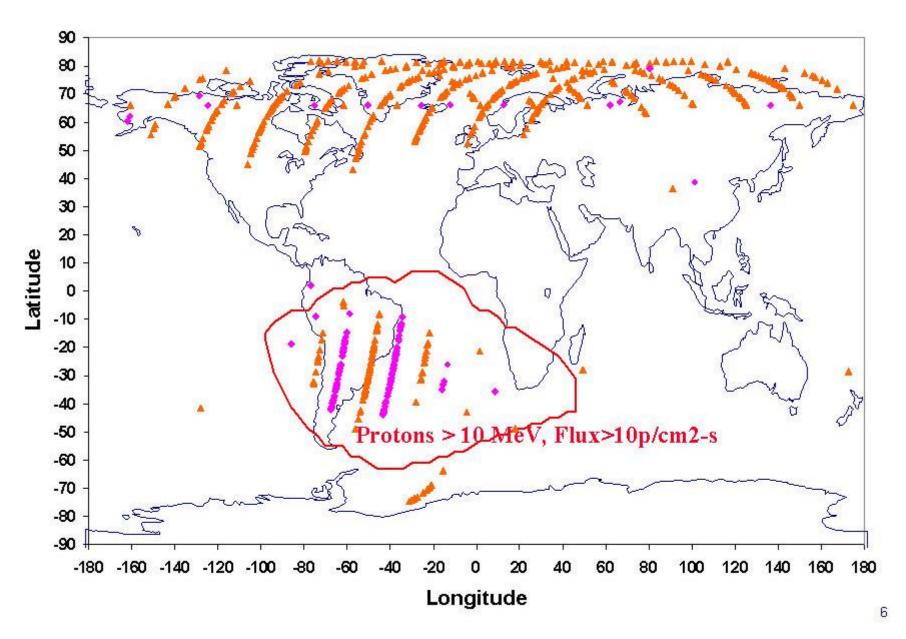
SeaStar Satellite: SEUs at 705 Km Altitude HIO-second transmissions only L January 1, 1999 - June 11, 2002



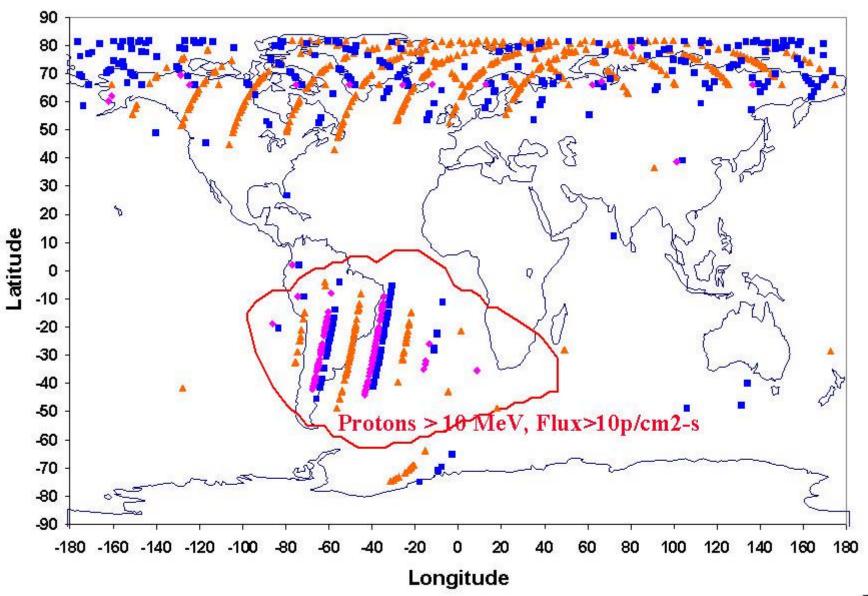
#### Typical day 7/13/200: ~ 80% of SEU occur in SAA



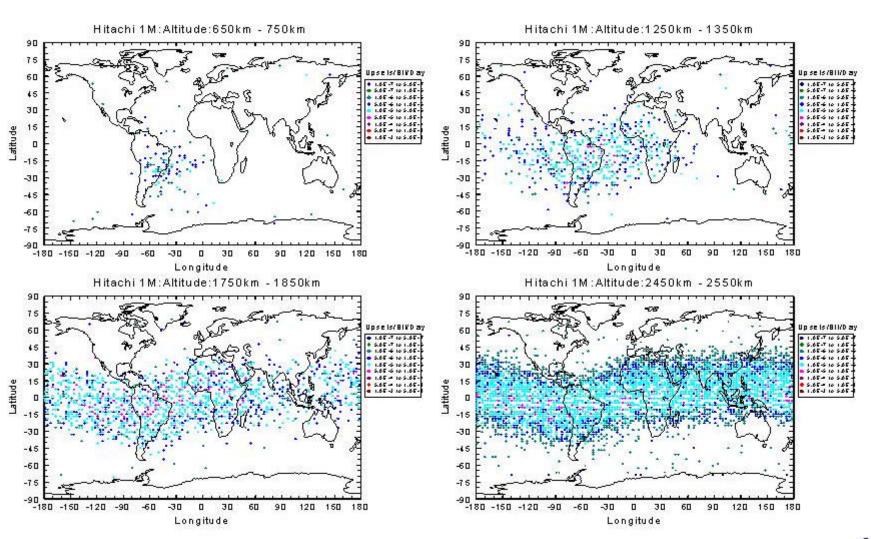
#### **Solar Event 7/14/2000**



#### Solar Event: 7/14-15/2000

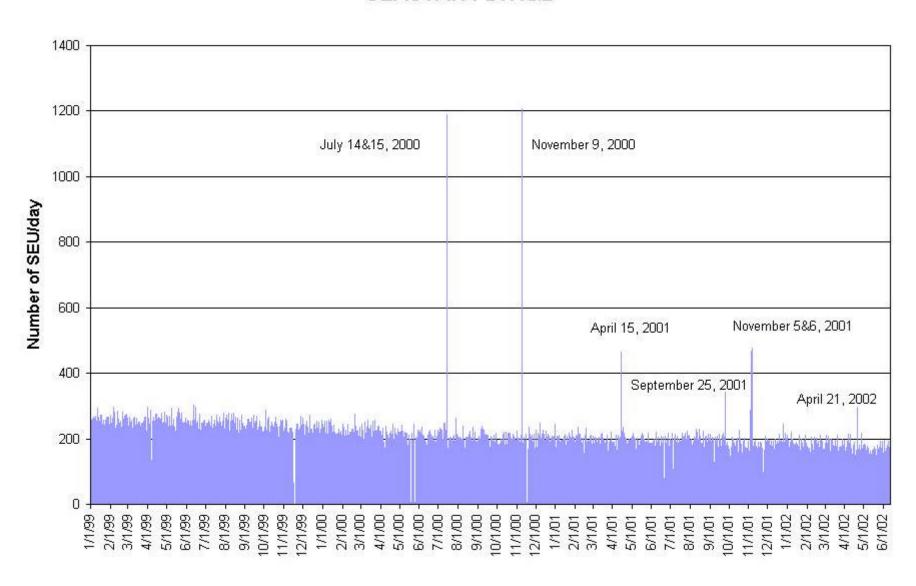


# SRAMs Upset Rates on CRUX/APEX

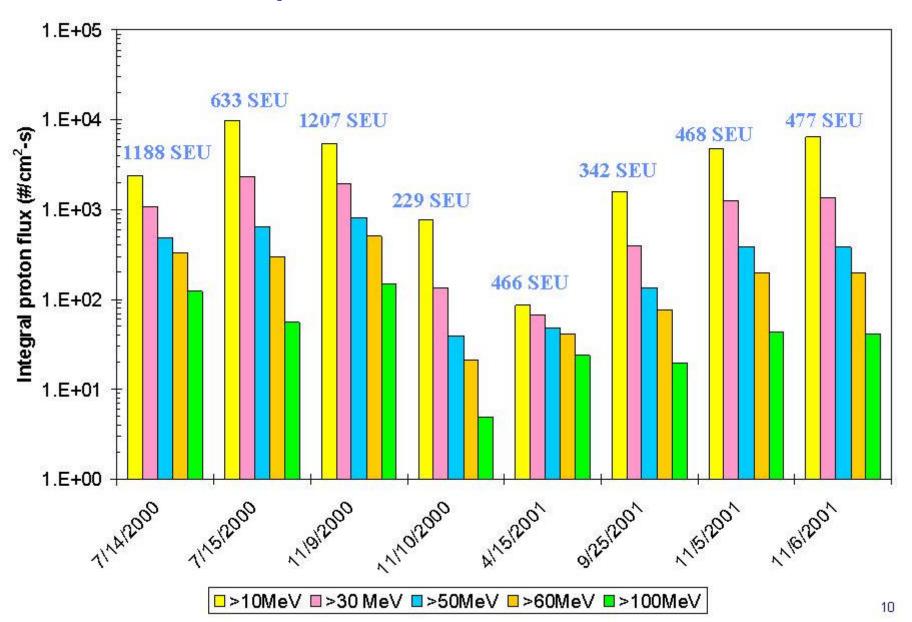


## Evolution of SEU rate versus time

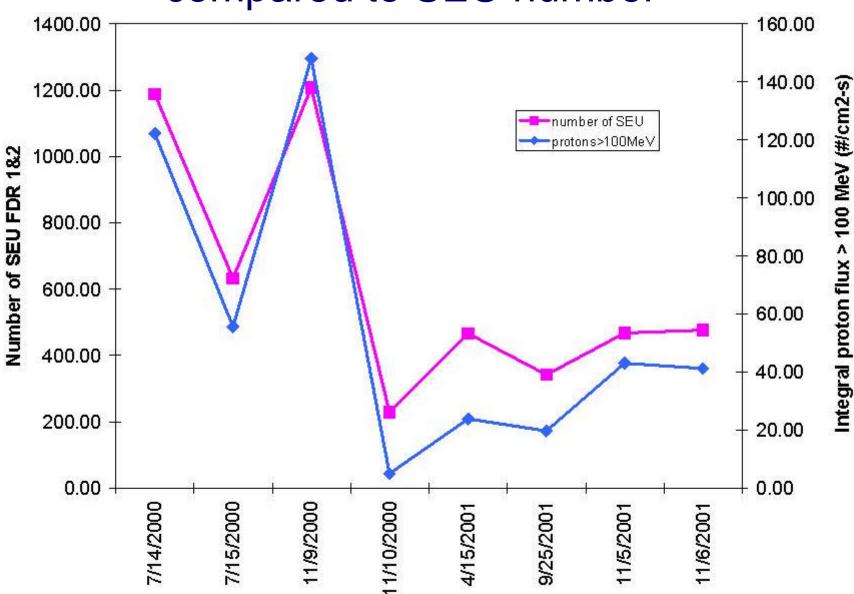
#### **SEASTAR FDR1&2**



## Proton composition of main solar events

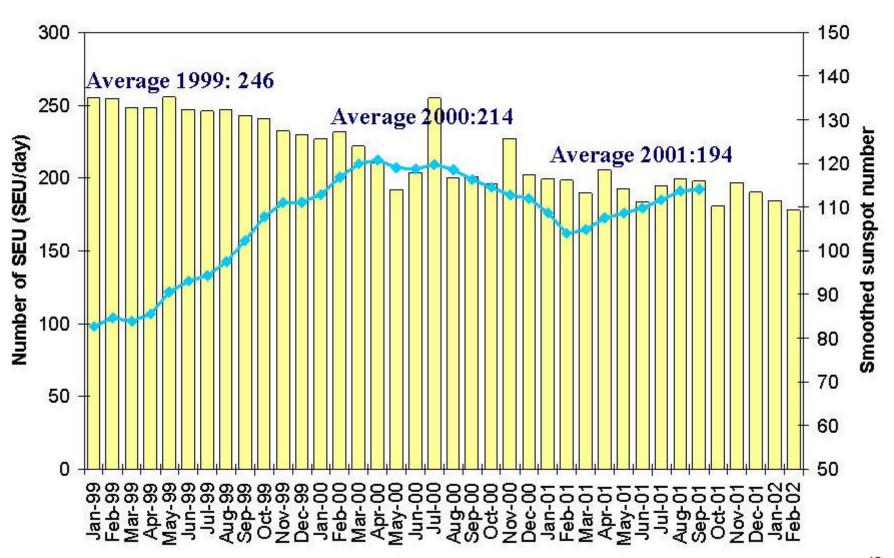


# Solar proton >100 MeV flux compared to SEU number



## Modulation of the SEU rate with solar activity

#### SEASTAR FDR1&2 monthly average



# Predictions of the upset rates

- Proton rates are generally underestimated and heavy ion rates overestimated, but generally an overall acceptable agreement is observed (within an order of magnitude) if:
  - Adequate environments models are used (CREME96) with the adequate solar modulation.
  - Heavy ion and proton data has been taken
- But discrepancies have been observed. They are due to:
  - SPE Environment models
  - Ground testing sample size
  - Assumptions on sensitive volume geometry for heavy ion rates

### SPE environment models

- SOHO SSR (DRAM 4M)
  - Predicted rate orders of magnitude higher than the observed rate during July 14 event (During quiet period agreement within a factor 2 with CREME96 model).
     Possible causes:
    - Shielding effect (assumed 1g/cm² of shielding)
    - CREME96 SPE models, (especially the ion component).
- MPTB DRAM (DRAM 16M) experiment
  - Predicted rates orders of magnitude higher than the observed rates during July 14 event (During quiet period, agreement within a factor 10). The cause has been attributed to the CREME96 model, because a good agreement was found with the actual measured spectra (CREDO3 instrument).

# **Ground Testing Sample Size**

- APEX CRUX (SRAM256K & 1M experiment), surprising results obtained because of large part to part variation (up to a factor 10)
  - Large part to part variation confirmed by ground testing
  - Variation of +/30% can be explained by the shielding
- HST SSR (12 Gbit made with 1440 DRAM 16M) anomaly was not detected by initial ground testing
  - Anomaly due to a SEFI, low device event sensitivity (~6E-13 cm²/die)
  - Sensitivity confirmed during ground testing of 100 parts
- Large part to part dispersion also observed on UOSAT and EXEQ experiments.

## Assumptions on Sensitive Volume Geometry

#### EXEQ experiments

- A conservative assumption of the sensitive volume thickness (2 μm) leads to a factor 45 overestimation of the flight rates on a 16M DRAM.
- A probing of the sensitive volume with low energy ions, showed a sensitive volume thickness of about 7 μm.
- With this value the ratio calculated rate to flight rate is reduced to a factor 4.

# Efficiency of SEU mitigation Schemes

- Generally EDAC techniques in association with a continuous scrubbing to "wash" the SEU have been successful.
  - Hamming:
    - SSR: SAMPEX (SRAM256K), TOMS (SRAM 256K), SEASTAR (DRAM4M), XTE(SRAM1M), UOSAT, SOHO(DRAM16M)
    - OBC memory: UOSAT (SRAM)
  - Reed Solomon:
    - SSR: UOSAT(SRAM256K&1M), UOSAT12(SRAM4M), HST (DRAM16M)
  - TMR
    - OBC memory: UOSAT12 (SRAM4M)
      - 200% overhead of memory size
      - No significant advantage compared to a modified Hamming code capable of correcting two errors per word.

# Multiple Bit Upset (MBU)

- A good SEU mitigation design needs a good understanding and ground testing of the MBU.
  - MBU mechanisms.
    - Charge diffusion away from an ion.
      - several neighboring cells are affected.
      - A different number of cells is affected depending on the ion LET and ion angle of incidence.
    - ion strike in the control circuitry.
      - Block error, row-column error .
    - Ion track intersecting a number of memory cells.
      - Difficult to test at ground level.

## MBU in space

- MBU have been observed in flight in all experiments
  - Most MBU occur in high latitude regions (more MBU caused by heavy ions than protons).
  - Blocks errors and multiple errors induced by a ion at grazing incidence have been identified on MPTB/DRAM experiment.
  - Significant rate (>10% of total upset rate) observed on more modern memories (DRAM>16M, SRAM 4M), SAC-ICARE experiment.
  - Single word multiple bit upset (SMU) have been reported on SAC-ICARE and UOSAT12 experiments.
  - Unexpected number of uncorrectable errors on CASSINI SSR.
    - The cause is a bad design: a data word is made with several memory word that have neighboring cells.
    - The uncorrectable error rate is still acceptable for the mission.

## Scrubbing Rates

- Traditionally scrubbing rates are calculated on the basis of daily averaged SEU rates.
- Flight data has shown that SEU occurs in burst (very high upset rate during a short period of time).
  - when the spacecraft goes trough the radiation belts.
  - During a SPE.
- Scrubbing rates need to be calculated for a peak upset rates.

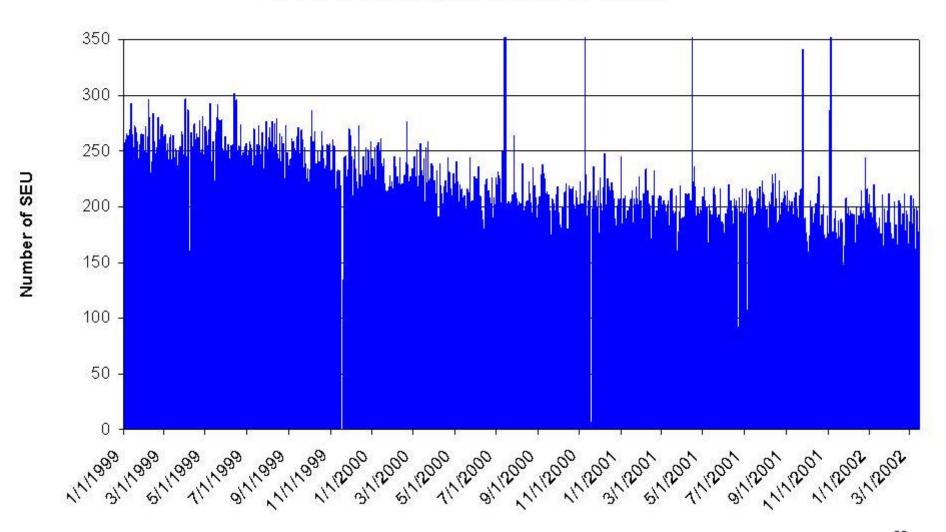
## Conclusion

- The use of high density commercial memories in space applications has been a success.
- Flight experiments have been very useful to understand the behavior of parts and validate the SEE mitigation schemes.
  - The experiments that include a radiation monitor allow a better understanding of the results (MPTB, SAC-ICARE).
  - It is also important to have an accurate information about the shielding thickness (MPTB, CRUX APEX).
- Modern memories are more and more complex (ie SDRAMs)
  - Significant sensitivity to SMU.
  - Significant SEFI sensitivity.
  - experiments are needed on these devices.

Back-up

## Modulation of the SEU rate with solar activity

#### SEASTAR FDR1&2, Number of SEU per Day



## SeaStar Mission

#### Altitude:

705 km - 705km

Inclination:

98.2°

Dates: September 1997 - Present

 Data from 1/1/1999 to 3/12/2002 has been analyzed for this study

#### Technology:

- FDR1&2 Seakr Solid State Recorders (SSR) w/ 64MB of Memory
  - EDAC (16,22) Modified Hamming Code - single bit correct, double bit detect
  - Telemetry gathered at 10 second intervals
  - Watchdog timer w/ soft reset 1 second timeout
- DRAM Hitachi MDM1400G-120, 4megabit x 1bit
  - 220 DRAMs per FDR
  - designed in 1994; now obsolete and hard to find

### Introduction

- SEASTAR in flight SEU performance were presented in SEE2000 symposium. We present an update that shows the effect of the solar activity and Solar Particle Events (SPE).
- Then we present the the anomaly on the Microwave Anisotropy Probe (MAP) that has been probably caused by solar heavy ions.

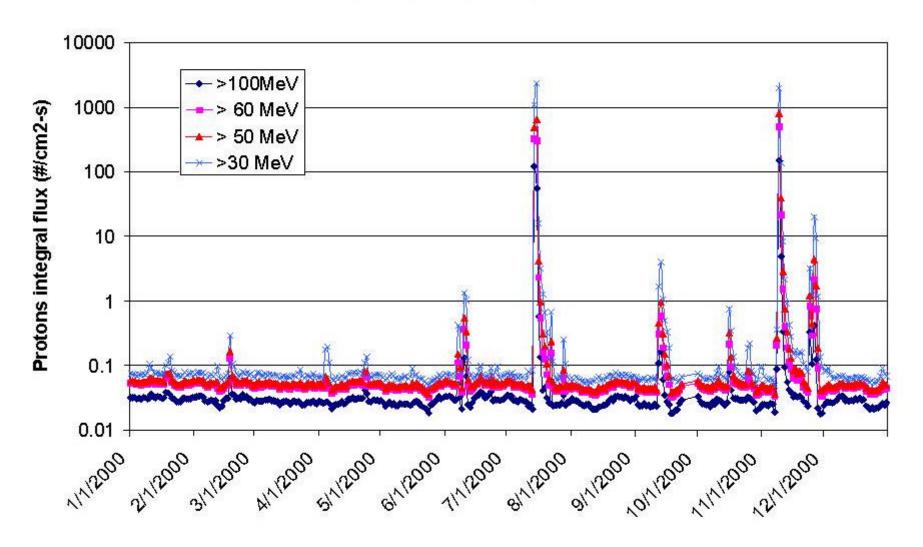
- SRAM
  - 0.5K
  - 1K
  - 4K
  - 64K
  - 256K
  - 1M
  - 4M

- DRAM
  - 4M
  - 16M
  - 64M

### Conclusion SEASTAR

- The data collected shows a significant sensitivity to solar protons.
- But only high energy protons (>100 MeV).
  This suggests a significant amount of shielding.
- The data also shows the fluctuation of the SEU numbers (and therefore the trapped proton fluxes) with the solar activity.

#### Solar protons flux - 2000



#### Solar protons 2001

